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May 5, 2000

BY FACSIMILE AND MAIL

David J. Cowan
3780 Wilshire Boulevard
Suite 910
Los Angeles CA 90010

Re: *Minovitch v. Battin* (LASC Case No. EC 224528)

Dear Mr. Cowan:

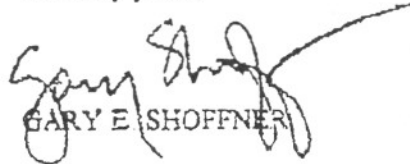
Attached is a copy of three pages of notes taken by then-student Walter Hollister during a March 21, 1960 class taught by Dr. Battin at MIT. The third page of the notes contains the following statement made by Dr. Battin during the lecture:

Possible to go to both Venus & Mars in 18 years in 1965 or so with about 16,000 ft/sec & return to earth.

I assume you and your client will understand the significance of Professor Hollister authenticating these notes as a contemporaneous record of Dr. Battin's statements to his Astronautical Guidance class in March 1960

I would like to suggest that your consultant also examine and test these notes while he is at the Draper Laboratory to examine the other materials. I understand Professor Hollister is willing to make them available for that purpose on reasonable notice. I have not talked to him, but I understand he can be reached at his office at (517) 253-2254.

Sincerely yours,


GARY E. SHOFFNER

$$\Delta v = 2v_0 \sin \delta$$



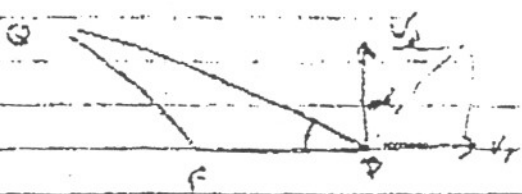
in rotating the relative velocity vector
 into that the initial velocity
 is changed in magnitude.

4. March 1960

Problems

#1 Let v_r & v_θ be the polar coordinate components of velocity of the vehicle at point P. Show that for the minimum energy path:

$$\frac{v_r}{v_\theta} = \tan\left(\frac{1}{2} \angle FPQ\right)$$



$$\alpha = \frac{1}{2} \angle QPF$$

#2 P 37 Eq. 2.2-8 factor of r_2 in denominator

$$v(r_1, \theta_1, \phi_1)$$

#3 Trajectory calculation in P. 219 is wrong
 had one slide rule

Round trip trajectory calculation

When T_0 there is one as if you go on a direct orbit
 two values of γ take in both to get there.
 trajectory

Round-trip

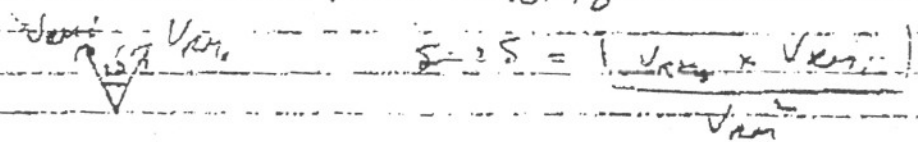
1. Assume a value for T_E , V_{RO} initial cond.
2. Guess at value of T_{EO} or fix the V_{RO} & add
3. Calculate a from T_E
4. Calculate c from a
5. Determine V_{RO} , V_{EO} , $V_{ROE} = V_{RO} - V_{EO}$
6. Compare computed V_{ROE} with assumed V_{RO} or adjust V_{RO}
7. If different change T_E & repeat
8. Calculate V_{SO} , V_{MO} & $V_{SM} = V_{SO} - V_{MO}$
 ↗ relative to Mars V_{SM}
9. Repeat 2-7 will

$T_E + T_E$ as time of launch from Mars

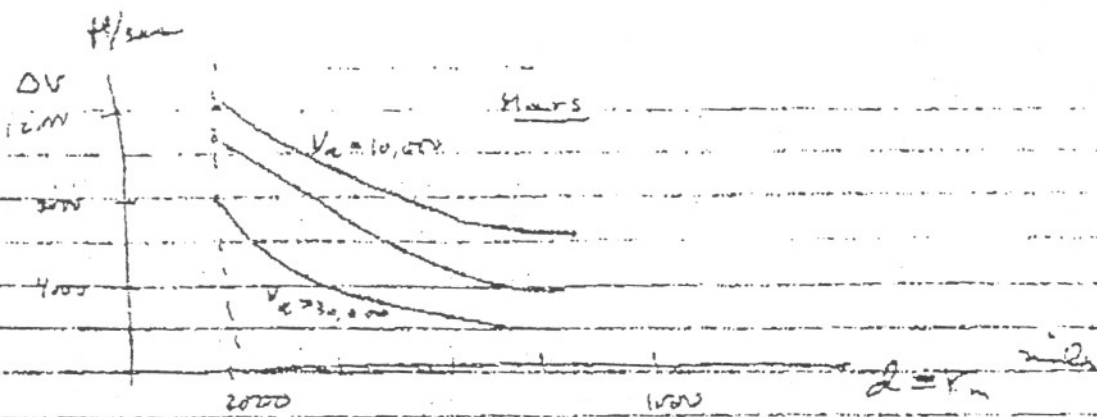
$|V_{SM}|$ as the departure velocity magnitude
 Mars only relative the relative velocity with

10. Inbound Velocity Vector V_{IM}

Outbound Velocity Vector V_{OM}



$$d = \frac{h_{M1}}{V_{SM}} (\csc \delta - 1)$$

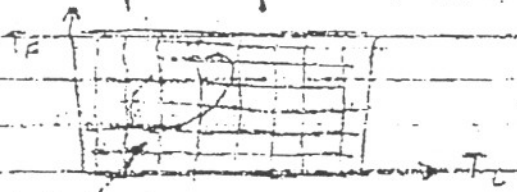


Propose to go to both Venus & Mars in 18 years in 1950s
with about 16,000 g/gal & return to earth

July Friday 24, 1950 Book a plane to describe the
Hawaii for a calculation

23 March

Describe all possible paths to Mars & Venus
as a function of T_e & T_c



Try this now on 7/24

Control of launch velocity or arrival velocity
Can then pick a time when conditions are most favorable

Still have to worry about where you pick it
(Injection from lower orbit might be preferable)

6 year periodicity of Earth, Venus & Mars

Synodic Period ≈ 0.5 ≈ 0.5

like First Passage

Time to adjust relative position 2.15 years ≈ 0.7
1.7 years ≈ 0.7
 ≈ 6 years ≈ 0.7